

تغيير شكل تير Deflection of Beams **تغییر شکل تیر**
Deflection of Beams
Care and Section 2014

رمتا مهالهى- دانشگاه آزلواسلامى داحد متران جنوب

روش انتگرال گيري – روش مستقيم

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Deformation of a Beam Under Transverse Loading

9 ‐ 3

سكاه آركواسلام بولعد ببرادن حنوب

• Relationship between bending moment and curvature for pure bending remains valid for general transverse loadings.

$$
\frac{1}{\rho} = \frac{M(x)}{EI}
$$

• Cantilever beam subjected to concentrated load at the free end,

$$
\rho \qquad EI
$$

Px

1

•Curvature varies linearly with x

• At the free end A,
$$
\frac{1}{\rho_A} = 0
$$
, $\rho_A = \infty$

• At the support *B*,
$$
\frac{1}{\rho_B} \neq 0
$$
, $|\rho_B| = \frac{EI}{PL}$

Deformation of a Beam Under Transverse Loading

- Overhanging beam
- Reactions at A and C
- Bending moment diagram
- Curvature is zero at points where the bending moment is zero, i.e., at each end and at E.

EI $\frac{1}{x} = \frac{M(x)}{x}$ ρ

- • Beam is concave upwards where the bending moment is positive and concave downwards where it is negative.
- • Maximum curvature occurs where the moment magnitude is a maximum.
- •An equation for the beam shape or *elastic curve* is required to determine maximum deflection and slope.

Equation of the Elastic Curve

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• From elementary calculus, simplified for beam parameters,

$$
\frac{1}{\rho} = \frac{d^2y}{\left[1 + \left(\frac{dy}{dx}\right)^2\right]^{3/2}} \approx \frac{d^2y}{dx^2}
$$

\n• Substituting and integrating
\n
$$
EI = EI \frac{d^2y}{dx^2} = M(x)
$$
\n
$$
EI \theta \approx EI \frac{dy}{dx} = \int_0^x M(x) dx + C
$$

• Substituting and integrating, $\frac{y}{2} = M(x)$ $\frac{1}{\rho} = EI \frac{d^2y}{dx^2} = M(x)$ $EI\frac{1}{\rho}$ = $EI\frac{d^2y}{dx^2}$ =

$$
EI \theta \approx EI \frac{dy}{dx} = \int_{0}^{x} M(x) dx + C_1
$$

$$
EI y = \int_{0}^{x} dx \int_{0}^{x} M(x) dx + C_1 x + C_2
$$

Equation of the Elastic Curve

• Constants are determined from boundary conditions

EI
$$
y = \int_{0}^{x} dx \int_{0}^{x} M(x) dx + C_1 x + C_2
$$

- \bullet Three cases for statically determinant beams,
	- Simply supported beam $y_A = 0$, $y_B = 0$
	- Overhanging beam $y_A = 0$, $y_B = 0$
	- Cantilever beam $y_A = 0$, $\theta_A = 0$
- More complicated loadings require multiple integrals and application of requirement for continuity of displacement and slope.

Direct Determination of the Elastic Curve From the Load Distribution

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 \bullet For a beam subjected to a distributed load,

$$
\frac{dM}{dx} = V(x) \qquad \frac{d^2M}{dx^2} = \frac{dV}{dx} = -w(x)
$$

•Equation for beam displacement becomes

$$
\frac{d^2M}{dx^2} = EI \frac{d^4y}{dx^4} = -w(x)
$$

- $EI y(x) = -\int dx \int dx \int dx \int w(x) dx$ $=-\int dx\int dx\int dx$ • Integrating four times yields
	- $3x + C_4$ $\frac{1}{2}C_2x^2$ $\frac{1}{6}C_1x^3 + \frac{1}{2}$ $+\frac{1}{6}C_1x^3+\frac{1}{2}C_2x^2+C_3x+C$
- • Constants are determined from boundary conditions.

Statically Indeterminate Beams

- Consider beam with fixed support at A and roller support at *B*.
- From free-body diagram, note that there are four unknown reaction components.
- Conditions for static equilibrium yield $\sum F_x = 0 \quad \sum F_y = 0 \quad \sum M_A = 0$

The beam is statically indeterminate.

•Also have the beam deflection equation,

$$
EI y = \int_{0}^{x} dx \int_{0}^{x} M(x) dx + C_1 x + C_2
$$

which introduces two unknowns but provides three additional equations from the boundary conditions:

At
$$
x = 0
$$
, $\theta = 0$ $y = 0$ At $x = L$, $y = 0$

For the prismatic beam and the loading shown (Fig. 9.16), determine the slope and deflection at point D .

We must divide the beam into two portions, AD and DB, and determine the function $y(x)$ which defines the elastic curve for each of these portions.

10 D ($x < L/4$). We draw the free-body diagram of
 $M_1 = \frac{3P}{4}x$ (9.17). Taking moments
 $M_1 = \frac{3P}{4}x$ (9.17)
 $\frac{1}{4}$

(9.18)
 $H = \frac{d^2y_1}{dx^2} = \frac{3}{4}Px$
 $\frac{dy_1}{dx^2} = \frac{3}{8}Px^2 + C_1$ (9.18)
 $H = \frac{P}{dx}$
 $\frac{d^2$

ر *متا مهلالهی۔ دانشگ*و آزلواسلامی داحد متر_لن جنوب

$$
EI\frac{d^2y_1}{dx^2} = \frac{3}{4}Px
$$

$$
EI \theta_1 = EI \frac{dy_1}{dx} = \frac{3}{8}Px^2 + C_1 \tag{9.19}
$$

$$
EI \, y_1 = \frac{1}{8}Px^3 + C_1x + C_2
$$

or, recalling Eq. (9.4) and rearranging terms,

$$
EI \frac{d^2 y_2}{dx^2} = -\frac{1}{4}Px + \frac{1}{4}PL \tag{9.22}
$$

$$
EI \theta_2 = EI \frac{dy_2}{dx} = -\frac{1}{8}Px^2 + \frac{1}{4}PLx + C_3 \tag{9.23}
$$

$$
EI y_2 = -\frac{1}{24}Px^3 + \frac{1}{8}PLx^2 + C_3x + C_4 \qquad (9.24)
$$

where $y_2(x)$ is the function which defines the el

of the beam. Integrating in x, we write
 $EI \theta_2 = EI \frac{dy_2}{dx} = -\frac{1}{B}Px^2 + \frac{1}{4}F$
 $EI \theta_2 = EI \frac{dy_2}{dx} = -\frac{1}{B}Px^2 + \frac{1}{4}F$
 $EI \theta_2 = -\frac{1}{24}Px^3 + \frac{1}{8}PLx^2 +$
 $\frac{1}{24}FLx$

$$
[x = 0, y_1 = 0], Eq. (9.20): \t 0 = C_2 \t (9.25)
$$

$$
[x = L, y_2 = 0]
$$
, Eq. (9.24):
$$
0 = \frac{1}{12}PL^3 + C_3L + C_4
$$
 (9.26)

$$
[x = L/4, \theta_1 = \theta_2]
$$
, Eqs. (9.19) and (9.23):

$$
\frac{3}{128}PL^2 + C_1 = \frac{7}{128}PL^2 + C_3 \tag{9.27}
$$

Fig. 9.19

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$$
[x = L/4, y_1 = y_2], \text{Eqs. (9.20) and (9.24):}
$$

$$
\frac{PL^3}{512} + C_1 \frac{L}{4} = \frac{11PL^3}{1536} + C_3 \frac{L}{4} + C_4
$$
(9.28)

$$
C_1 = -\frac{7PL^2}{128}, C_2 = 0, C_3 = -\frac{11PL^2}{128}, C_4 = \frac{PL^3}{384}
$$

$$
EI \theta_1 = \frac{3}{8}Px^2 - \frac{7PL^2}{128} \tag{9.29}
$$

$$
EI\ y_1 = \frac{1}{8}Px^3 - \frac{7PL^2}{128}x\tag{9.30}
$$

 $\frac{1}{512} + C_1 \frac{1}{4} = \frac{1}{1536} + C_3 \frac{1}{4}$

Solving these equations simultaneously, we find
 $C_1 = -\frac{7PL^2}{128}$, $C_2 = 0$, $C_3 = -\frac{11PL^2}{128}$,

Substituting for C_1 and C_2 into Eqs. (9.19) and (9.
 $x \le L/4$,
 EI

$$
\theta_D = -\frac{PL^2}{32EI} \quad \text{and} \quad y_D = -\frac{3PL^3}{256EI}
$$

 $P = 50$ kips $L = 15$ ft $a = 4$ ft $W14 \times 68$ $I = 723 \text{ in}^4$ $E = 29 \times 10^6 \text{ psi}$

For portion AB of the overhanging beam, (a) derive the equation for the elastic curve, (b) determine the maximum deflection,

 (c) evaluate y_{max} .

<mark>0</mark> ارگ^{ونس}لام ہوگئے ہیں ان حوب²

SOLUTION:

- Develop an expression for $M(x)$ and derive differential equation for elastic curve.
- • Integrate differential equation twice and apply boundary conditions to obtain elastic curve.
- Locate point of zero slope or point of maximum deflection.
- Evaluate corresponding maximum deflection.

SOLUTION:

• Develop an expression for M(x) and derive differential equation for elastic curve.

- Reactions:

$$
R_A = \frac{Pa}{L} \downarrow \quad R_B = P\left(1 + \frac{a}{L}\right)\uparrow
$$

- From the free-body diagram for section AD, $M = -P\frac{a}{L}x \quad (0 < x < L)$
- - The differential equation for the elastic curve,

$$
EI\frac{d^2y}{dx^2} = -P\frac{a}{L}x
$$

 $\frac{d^2y}{dx^2} = -P\frac{d^2y}{dx^2}$

 $EI\frac{d^2y}{dx^2} = \frac{2y}{2}$

x L

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• Integrate differential equation twice and apply boundary conditions to obtain elastic curve.

$$
E I_x = L, y = 0
$$
 boundary conditions to obtain
\n
$$
E I \frac{dy}{dx} = -\frac{1}{2} P \frac{a}{L} x^2 + C_1
$$
\n
$$
E I y = -\frac{1}{6} P \frac{a}{L} x^3 + C_1 x + C_2
$$
\nat $x = 0, y = 0$: $C_2 = 0$
\n $P \frac{a}{L} x$
\nat $x = L, y = 0$: $0 = -\frac{1}{6} P \frac{a}{L} L^3$
\nSubstituting,
\n
$$
E I \frac{dy}{dx} = -\frac{1}{2} P \frac{a}{L} x^2 + \frac{1}{6} P a L
$$

at
$$
x = 0
$$
, $y = 0$: $C_2 = 0$

at
$$
x = L
$$
, $y = 0$: $0 = -\frac{1}{6}P\frac{a}{L}L^3 + C_1L$ $C_1 = \frac{1}{6}PaL$

Substituting,

$$
EI\frac{dy}{dx} = -\frac{1}{2}P\frac{a}{L}x^2 + \frac{1}{6}PaL \quad \frac{dy}{dx} = \frac{Pal}{6EI} \left[1 - 3\left(\frac{x}{L}\right)^2\right]
$$

$$
EI y = -\frac{1}{6} P \frac{a}{L} x^3 + \frac{1}{6} PaL
$$

$$
EI y = -\frac{1}{6} P \frac{a}{L} x^3 + \frac{1}{6} P a L x
$$

$$
y = \frac{P a L^2}{6EI} \left[\frac{x}{L} - \left(\frac{x}{L}\right)^3 \right]
$$

$$
y = \frac{PaL^2}{6EI} \left[\frac{x}{L} - \left(\frac{x}{L}\right)^3 \right]
$$

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• Locate point of zero slope or point of maximum deflection.

$$
\frac{dy}{dx} = 0 = \frac{Pal}{6EI} \left[1 - 3 \left(\frac{x_m}{L} \right)^2 \right] \quad x_m = \frac{L}{\sqrt{3}} = 0.577L
$$

• Evaluate corresponding maximum deflection.

$$
y_{\text{max}} = \frac{PaL^2}{6EI} \left[0.577 - (0.577)^3 \right]
$$

$$
y_{\text{max}} = 0.0642 \frac{PaL^2}{6EI}
$$

$$
y_{\text{max}} = 0.0642 \frac{(50 \text{kips})(48 \text{ in})(180 \text{ in})^2}{6(29 \times 10^6 \text{ psi})(723 \text{ in}^4)}
$$

$$
y_{\text{max}} = 0.238 \text{ in}
$$

For the uniform beam, determine the reaction at A, derive the equation for the elastic curve, and determine the slope at A. (Note that the beam is statically indeterminate to the first degree) **Jones Concernsity**
 B
 B
 B
 Develop the differe

the elastic curve (we dependent on the relastic curve (we dependent on the relastic curve, and

peat *A*. (Note that be slope and by indeterminate)

 Jones Condi

SOLUTION:

- Develop the differential equation for the elastic curve (will be functionally dependent on the reaction at A).
- • Integrate twice and apply boundary conditions to solve for reaction at A and to obtain the elastic curve.
- Evaluate the slope at A.

• Consider moment acting at section D,

$$
\sum_{y} w = w_0 \frac{x}{L}
$$

\n
$$
R_A x - \frac{1}{2} \left(\frac{w_0 x^2}{L} \right) \frac{x}{3} - M = 0
$$

\n
$$
M = R_A x - \frac{w_0 x^3}{6L}
$$

\n• The differential equation for
\ncurve,
\n
$$
EI \frac{d^2 y}{dx^2} = M = R_A x - \frac{w_0 x^3}{6L}
$$

• The differential equation for the elastic curve,

$$
EI\frac{d^2y}{dx^2} = M = R_A x - \frac{w_0 x^3}{6L}
$$

$$
EI\frac{d^2y}{dx^2} = M = R_A x - \frac{w_0 x^3}{6L}
$$

رەيمامىلالىق-دا*ڭگاھ آزل*واسلامى داھدىتىرن جۇپ¹⁸ -

• Integrate twice

$$
[x = L, \theta = 0]
$$

\n
$$
[x = L, y = 0]
$$

\n
$$
E I \frac{dy}{dx} = E I \theta = \frac{1}{2} R_A x^2 - \frac{w_0 x^4}{24L} + C_1
$$

\n
$$
E I y = \frac{1}{6} R_A x^3 - \frac{w_0 x^5}{120L} + C_1 x + C_2
$$

\n• Apply boundary conditions:
\n
$$
A x - \frac{w_0 x^3}{6L}
$$

\nat $x = 0, y = 0$: $C_2 = 0$
\nat $x = L, \theta = 0$: $\frac{1}{2} R_A L^2 - \frac{w_0 L^3}{24} + C_2$
\nat $x = L, y = 0$: $\frac{1}{6} R_A L^3 - \frac{w_0 L^4}{120} + C_2$
\n• Solve for reaction at A

• Apply boundary conditions:

at
$$
x = 0
$$
, $y = 0$: $C_2 = 0$

at
$$
x = L
$$
, $\theta = 0$: $\frac{1}{2}R_A L^2 - \frac{w_0 L^3}{24} + C_1 = 0$
at $x = L$, $y = 0$: $\frac{1}{6}R_A L^3 - \frac{w_0 L^4}{120} + C_1 L + C_2 = 0$

•Solve for reaction at A

$$
\frac{1}{3}R_A L^3 - \frac{1}{30} w_0 L^4 = 0
$$

$$
R_A = \frac{1}{10} w_0 L \uparrow
$$

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• Substitute for C_1 , C_2 , and R_A in the elastic curve equation,

$$
EI y = \frac{1}{6} \left(\frac{1}{10} w_0 L \right) x^3 - \frac{w_0 x^5}{120L} - \left(\frac{1}{120} w_0 L^3 \right) x
$$

$$
y = \frac{w_0}{120ELL} \left(-x^5 + 2L^2 x^3 - L^4 x \right)
$$

Differentiate once to find the slope,

$$
\theta = \frac{dy}{dx} = \frac{w_0}{120EIL} \left(-5x^4 + 6L^2x^2 - L^4 \right)
$$

at
$$
x = 0
$$
,
$$
\theta_A = \frac{w_0 L^3}{120EI}
$$

•

EXAMPLE 9.04

The simply supported prismatic beam AB carries a uniformly distributed load w per unit length (Fig. 9.21). Determine the equation of the elastic curve and the maximum deflection of the beam. (This is the same beam and loading as in Example 9.02.)

$$
EI \frac{d^4 y}{dx^4} = -w
$$

\n
$$
EI \frac{d^3 y}{dx^3} = V(x) = -wx + C_1
$$

\n
$$
EI \frac{d^2 y}{dx^2} = M(x) = -\frac{1}{2}wx^2 + C_1x + C_2
$$
 (9.34)

$$
EI \frac{d^3y}{dx^4} = -w
$$

\n
$$
EI \frac{d^3y}{dx^3} = V(x) = -wx + C_1
$$

\n
$$
EI \frac{d^3y}{dx^3} = V(x) = -wx + C_1
$$

\n
$$
EI \frac{d^3y}{dx^3} = W(x) = -\frac{1}{2}wx^2 + C_1x + C_2
$$
 (9.34)
\nNoting that the boundary conditions require that $M = 0$ at both ends of the beam (Fig. 9.22), we first let $x = 0$ and $M = 0$ in Eq. (9.34) and obtain $C_2 = 0$. We then make $x = L$ and $M = 0$ in the same equation and obtain $C_1 = \frac{1}{2}wL$.
\nCarrying the values of C_1 and C_2 back into Eq. (9.34), and integration in the surface of C_1 and C_2 back into Eq. (9.34), and integration in the direction $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M = 0$ and $M = 0$ in the same equation $x = L$, $M =$

Fig. 9.22

 $x = 0, y = 0$

ر <mark>شام م</mark>لایس<u>ی- دانشگو آزلواسلامی داند. تهرن</u> جنوبه

But the boundary conditions also require that $y = 0$ a

beam. Letting $x = 0$ and $y = 0$ in Eq. (9.35), we obtain
 $0 = -\frac{1}{24}wL^4 + \frac{1}{12}wL^4 + C_3L$
 $C_3 = -\frac{1}{24}wL^3$

Carrying the values of C_3 and C_4 back into

$$
0 = -\frac{1}{24}wL^4 + \frac{1}{12}wL^4 + C_3L
$$

$$
C_3 = -\frac{1}{24}wL^3
$$

$$
y = \frac{w}{24EI}(-x^4 + 2Lx^3 - L^3x) \tag{9.36}
$$

$$
y|_{\text{max}} = \frac{5wL^4}{384EI}
$$

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تحلیل تیرهای نامعین
اده از معادلات تغییر مکان تیر
روش نیرو
المالی **تحليل تيرهاي نامعين با استفاده از معادلات تغيير مكان تير روش نيرو** رمتا مىلالىق- دانشگاه آزلواسلامى داھدىتىرن جونب

Method of Superposition

Principle of Superposition:

- Deformations of beams subjected to combinations of loadings may be obtained as the linear combination of the deformations from the individual loadings
- \bullet Procedure is facilitated by tables of solutions for common types of loadings and supports.

For the beam and loading shown, determine the slope and deflection at point B.

SOLUTION:

Superpose the deformations due to *Loading I* and *Loading II* as shown.

Loading I

$$
(\theta_B)_I = -\frac{wL^3}{6EI} \qquad (y_B)_I = -\frac{wL^4}{8EI}
$$

Loading II

 $(\theta_C)_{II} = \frac{WL}{48EI}$ wL^3 $(y_C)_{II} = \frac{wL^4}{128EI}$

In beam segment CB, the bending moment is zero and the elastic curve is a straight line.

$$
(\theta_B)_{II} = (\theta_C)_{II} = \frac{wL^3}{48EI}
$$

$$
(y_B)_{II} = \frac{wL^4}{128EI} + \frac{wL^3}{48EI} \left(\frac{L}{2}\right) = \frac{7wL^4}{384EI}
$$

Combine the two solutions,

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$$
\theta_B = (\theta_B)_I + (\theta_B)_H = -\frac{wL^3}{6EI} + \frac{wL^3}{48EI}
$$
\n
$$
\theta_B = \frac{7wL^3}{48EI}
$$

$$
y_B = (y_B)_I + (y_B)_H = -\frac{wL^4}{8EI} + \frac{7wL^4}{384EI}
$$
\n
$$
y_B = \frac{41wL^4}{384EI}
$$

Application of Superposition to Statically Indeterminate Beams

- \bullet Method of superposition may be applied to determine the reactions at the supports of statically indeterminate beams.
- • Designate one of the reactions as redundant and eliminate or modify the support.

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- • Determine the beam deformation without the redundant support.
- • Treat the redundant reaction as an unknown load which, together with the other loads, must produce deformations compatible with the original supports.

For the uniform beam and loading shown, determine the reaction at each support and the slope at end A.

SOLUTION:

- Release the "redundant" support at B, and find deformation.
- \bullet Apply reaction at B as an unknown load to force zero displacement at B .

• Distributed Loading:

$$
(y_B)_w = -\frac{w}{24EI} \left[\left(\frac{2}{3}L\right)^4 - 2L\left(\frac{2}{3}L\right)^3 + L^3\left(\frac{2}{3}L\right) \right]
$$

\n= -0.01132 $\frac{wL^4}{EI}$
\n
\n
$$
(y_B)_w
$$

\n
$$
(y_B)_R = \frac{R_B}{3EIL} \left(\frac{2}{3}L\right)^2 \left(\frac{L}{3}\right)^2 = 0.01646 \frac{R_B L^3}{EI}
$$

\n
\nFor compatibility with original supports, y_B
\n
$$
0 = (y_B)_w + (y_B)_R = -0.01132 \frac{wL^4}{EI} + 0.01646 \frac{R_B}{EI}
$$

• Redundant Reaction Loading:

$$
(y_B)_R = \frac{R_B}{3EIL} \left(\frac{2}{3}L\right)^2 \left(\frac{L}{3}\right)^2 = 0.01646 \frac{R_B L^3}{EI}
$$

- •For compatibility with original supports, y_B = $0 = (y_B)_w + (y_B)_R = -0.01132 \frac{wL^4}{EI} + 0.01646 \frac{R_B L^3}{EI}$ $R_B = 0.688$ wL \uparrow
- •From statics,

$$
R_A = 0.271 wL \uparrow \qquad R_C = 0.0413 wL \uparrow
$$

Slope at end A,

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$$
(\theta_A)_w = -\frac{wL^3}{24EI} = -0.04167 \frac{wL^3}{EI}
$$

$$
(\theta_A)_R = \frac{0.0688 \, wL}{6EIL} \left(\frac{L}{3}\right) \left[L^2 - \left(\frac{L}{3}\right)^2\right] = 0.03398 \, \frac{wL^3}{EI}
$$

$$
\theta_A = (\theta_A)_w + (\theta_A)_R = -0.04167 \frac{wL^3}{EI} + 0.03398 \frac{wL^3}{EI} \qquad \theta_A = -0.00769 \frac{wL^3}{EI}
$$

$$
\theta_A = -0.00769 \frac{wL^3}{EI}
$$

Moment‐Area Theorems

 $\frac{M}{E1}$ \overline{C}

داسکاه آراداسلامی بالحد بهران حوب²

- Geometric properties of the elastic curve can be used to determine deflection and slope.
- Consider a beam subjected to arbitrary loading,

be used to determine deflection and s
\n
$$
D \rightarrow R
$$
\nConsider a beam subjected to arbitrary
\n
$$
\frac{d\theta}{dx} = \frac{d^2y}{dx^2} = \frac{M}{EI}
$$
\n
$$
d\theta = \frac{M}{EI} dx
$$
\n
$$
-D \rightarrow \theta_C
$$
\n
$$
D \theta_C
$$
\n
$$
\theta_D - \theta_C = \int_{x_C}^{x_D} \frac{M}{EI} dx
$$
\n
$$
\theta_D - \theta_C = \int_{x_C}^{x_D} \frac{M}{EI} dx
$$
\n
$$
\theta_D - \theta_C = \int_{x_C}^{x_D} \frac{M}{EI} dx
$$
\n
$$
\theta_D - \theta_C = \int_{x_C}^{x_D} \frac{M}{EI} dx
$$

$$
\int_{\theta_C}^{\theta_D} d\theta = \int_{x_C}^{\theta_D} \frac{M}{EI} dx
$$

$$
\theta_D - \theta_C = \int_{x_C}^{x_D} \frac{M}{EI} \, dx
$$

•First Moment-Area Theorem:

> $\theta_{D/C}$ = area under *(M/EI)* diagram between C and D.

Moment‐Area Theorems

•Tangents to the elastic curve at P and P' intercept a segment of length dt on the vertical through C .

•

 $=$ tangential deviation of C with respect to D

Second Moment-Area Theorem:

The tangential deviation of C with respect to D is equal to the first moment with respect to a vertical axis through C of the area under the (M/EI) diagram between C and D.

Application to Cantilever Beams and Beams With Symmetric Loadings

•

• Cantilever beam - Select tangent at A as the reference.

 $\sqrt{ }$ $\theta_D = \theta_{D/C}$ $t_{D/C}$ Reference tangent منتخلا از دان اسلامی دان سران حوب⁴

 Simply supported, symmetrically loaded beam - select tangent at C as the reference.

$$
y_D = t_{D/C} - t_{B/C}.
$$

Bending Moment Diagrams by Parts

- Determination of the change of slope and the tangential deviation is simplified if the effect of each load is evaluated separately.
- Construct a separate (M/EI) diagram for each load.
	- The change of slope, $\theta_{D/C}$ is obtained by adding the areas under the diagrams.
	- The tangential deviation, $t_{D/C}$ is obtained by adding the first moments of the areas with respect to a vertical axis through D.
- Bending moment diagram constructed from individual loads is said to be *drawn by parts*.
Sample Problem 9.11

For the prismatic beam shown, determine the slope and deflection at E .

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SOLUTION:

- Determine the reactions at supports.
- \bullet Construct shear, bending moment and (M/EI) diagrams.
- •Taking the tangent at C as the reference, evaluate the slope and tangential deviations at E.

Sample Problem 9.11

SOLUTION:

 \bullet Determine the reactions at supports.

$$
R_B = R_D = wa
$$

• Construct shear, bending moment and (M/EI) diagrams.

$$
A_1 = -\frac{wa^2}{2EI} \left(\frac{L}{2}\right) = -\frac{wa^2L}{4EI}
$$

$$
A_2 = -\frac{1}{3} \left(\frac{wa^2}{2EI}\right)(a) = -\frac{wa^3}{6EI}
$$

Sample Problem 9.11

Slope at E:
\n
$$
\theta_E = \theta_C + \theta_{E/C} = \theta_{E/C}
$$
\n
$$
= A_1 + A_2 = -\frac{wa^2L}{4EI} - \frac{wa^3}{6EI}
$$
\n
$$
\theta_E = -\frac{wa^2}{12EI}(3L + 2a)
$$

•

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 $=\left[-\frac{wa^{3}L}{4EI}-\frac{wa^{2}L^{2}}{16EI}-\frac{wa^{4}}{8EI}\right]-\left[-\frac{wa^{2}L^{2}}{16EI}\right]$ $=\left[A_1\left(a+\frac{L}{4}\right)+A_2\left(\frac{3a}{4}\right)\right]-\left[A_1\left(\frac{L}{4}\right)\right]$ $y_E = t_{E/C} - t_{D/C}$ *EI Lwa EI wa EI Lwa EI Lwa* $A_1\left(a + \frac{L}{a}\right) + A_2\left(\frac{3a}{a}\right)\Big| - \Big|A_1\Big|$ 4*EI* 16*EI* 8*EI* | 16 4) | ' (4 3 4 $3₁$ $2₁2$ $4₁$ $2₁2$ $|1|$ $u + \frac{1}{4}$ $| + A2|$ $\frac{1}{4}$ $| - |A1|$ Deflection at E: $\theta_E = \theta_C + \theta_{E/C} = \theta_{E/C}$
 $\frac{1}{\theta_E}$
 $\theta_E = -\frac{wa^2L}{4EI}$
 $\theta_E = -\frac{wa^2}{12EI}(3L + 2a)$
 $\theta_E = -\frac{wa^2L}{12EI}(3L + 2a)$
 $\theta_E = -\frac{wa^2L}{12EI}(3L + 2a)$

$$
y_E = -\frac{wa^3}{8EI}(2L+a)
$$

Application of Moment‐Area Theorems to Beams With Unsymmetric Loadings

• Define reference tangent at support A. Evaluate $\theta_{\!A}$ by determining the tangential deviation at B with respect to A.

$$
\theta_{A}=-\frac{t_{B/A}}{L}
$$

• The slope at other points is found with respect to reference tangent.

 $\theta_D = \theta_A + \theta_{D/A}$

•The deflection at D is found from the tangential deviation at *D*.

$$
\frac{EF}{x} = \frac{HB}{L} \qquad \text{or} \qquad EF = \frac{x}{L} t_{B/A}
$$

$$
y_D = ED - EF = t_{D/A} - \frac{x}{L} t_{B/A}
$$

Maximum Deflection

• Maximum deflection occurs at point K where the tangent is horizontal.

$$
\theta_{K/A} = \theta_K - \theta_A = 0 - \theta_A = -\theta_A
$$

- •Point *K* may be determined by measuring an area under the (M/EI) diagram equal to - $\theta_{\!A}$.
- Obtain y_{max} by computing the first moment with respect to the vertical axis through A of the area between A and K .

Use of Moment‐Area Theorems With Statically Indeterminate Beams

- Reactions at supports of statically indeterminate beams are found by designating a redundant constraint and treating it as an unknown load which satisfies a displacement compatibility requirement.
- The (M/EI) diagram is drawn by parts. The resulting tangential deviations are superposed and related by the compatibility requirement.
- \bullet With reactions determined, the slope and deflection are found from the moment-area method.

6.6 Conjugate-Beam Method

The conjugate-beam method, developed by Otto Mohr in 1868, generally

provides a more convention means of comparation of beams than the moment-area method. Althoughta putational effort required by the two methods is conjugate-beam method is preferred by many systematic sign convention and st

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 $\frac{dS}{dx} = w$ $\frac{d^2M}{dx^2} = w$ $\frac{d^2y}{dx^2} = \theta$ or $\frac{d^2y}{dx^2} =$
performed to compute shear and bending moment, respective
load. Furthermore, if the *M/EI* diagram for a beam is applied
on a fictitious analogous beam, t

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Supports for Conjugate Beams

External supports and internal connections for conjugate beams are deterd from the analogous relationships between conjugate sponding real beams; that is, the shear and bending t on the conjugate beam must be consistent with th on at that point on the real beam. The conjugate cous types of rea slope is continuous (i.e., there is no abrupt change of slope from one side

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FIG. 6.13

 (h)

 (i)

 $\langle j \rangle$

≞

∸

Procedure for Analysis

The following step-by-step procedure can be used for determining the slopes and deflections of beams by the conjugate-beam method.

- 1. Construct the M/EI diagram for the given (real) beam subjected to the specified (real) loading. If the beam is subjected to
-
-
-
- a combination of different types of loads (e.g., conductional conduction be capplied by constructing the M/EI diagram by parametering sections and interprecial process and interprecial be beam. The external supports and i
	-
	- ment at that point on the conjugate beam. A positive bending moment in the conjugate beam denotes a positive or upward deflection of the real beam and vice versa.

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Example 6.8

and deflections at points *B* and *C* of the cantilever beam shown in Fig. 6.14(a) beam was analyzed in Example 6.3 by the moment-area method. The M/EI is $I = 3,000$ in ⁴ is shown in Fig. 6.14(b). beam method.

Solution

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$$
+ \uparrow S_B = \frac{1}{EI} \left[-100(15) - \frac{1}{2}(150)(15) \right] = -\frac{2,625 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
B = -\frac{2.625 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
\theta_B = -\frac{2,625(12)^2}{(29,000)(3,000)} = -0.0043 \text{ rad}
$$

\n
$$
\theta_B = 0.0043 \text{ rad}
$$

Deflection at B . The deflection at B on the real beam is equal to the bending moment at B in the conjugate beam. Using the free body of the conjugate beam to the left of B and considering the clockwise moments of the external forces about

$$
\sum_{i=1}^{n} \alpha_i
$$

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 B as positive, in accordance with the beam sign convention (Fig. 5.2), we compute the bending moment at B on the conjugate beam as

$$
+\zeta M_B = \frac{1}{EI} \left[-100(15)(7.5) - \frac{1}{2}(150)(15)(10) \right] = -\frac{22,500 \text{ k} \cdot \text{ft}^3}{EI}
$$

Therefore, the deflection at B on the real beam is

+
$$
\uparrow
$$
 S_C = $\frac{1}{EI} \left[-100(15) - \frac{1}{2}(150)(15) - \frac{1}{2}(200)(10) \right] = -\frac{3{,}625 \text{ k} \cdot \text{ft}^2}{EI}$

$$
\theta_C = -\frac{3{,}625 \text{ k-fr}}{EI} = -\frac{3{,}625(12)^2}{(29{,}000)(3{,}000)} = -0.006 \text{ rad}
$$

$$
\theta_C = 0.006 \text{ rad}
$$

$$
\Delta_B = 0.45 \text{ in. } 1
$$

\nSlope at *C*. Using the free body of the conjugate beam to the left of *C*, we determine the shear
\n
$$
+ \int S_C = \frac{1}{EI} \left[-100(15) - \frac{1}{2}(150)(15) - \frac{1}{2}(200)(10) \right] = -\frac{3,625 \text{ k} \cdot \text{ft}^2}{EI}
$$

\nTherefore, the slope at *C* on the real beam is
\n
$$
\theta_C = -\frac{3,625 \text{ k} \cdot \text{ft}^2}{EI} = -\frac{3,625(12)^2}{(29,000)(3,000)} = -0.006 \text{ rad}
$$

\n
$$
\theta_C = 0.006 \text{ rad}
$$

\nDeflection at *C*. Considering the free body of the conjugate beam to the left of *C*, we obtain
\n
$$
+ \left\langle M_C = \frac{1}{EI} \left[-100(15)(17.5) - \frac{1}{2}(150)(15)(20) - \frac{1}{2}(200)(10)(6.67) \right] = -\frac{55,420 \text{ k} \cdot \text{ft}^3}{EI} = -\frac{55,420 \text{ k} \cdot \text{ft}^3}{(29,000)(3,000)} = -1.1 \text{ in.}
$$

\n
$$
\Delta_C = 1.1 \text{ in.} \downarrow
$$

\nExample 6.9
\nDetermine the slope and deflection at point *B* of the beam shown in Fig. 6.15(a) by the conjugate beam.
\nSolution
\n*MIEI* Diagram. See Fig. 6.15(b).
\nConjugate beam. The conjugate beam, loaded with the *M/EI* diagram of the real beam, is show

$$
\Delta_C = -\frac{55,420 \text{ k} \cdot \text{ft}^3}{EI} = -\frac{55,420(12)^3}{(29,000)(3,000)} = -1.1 \text{ in.}
$$

$$
\Delta_C = 1.1 \text{ in.} \downarrow
$$

Ans.

Ans.

Ans.

 $\frac{1}{2}$

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Slope at B . Considering the free body of the conjugate beam to the left of B , we determine the shear at B as

$$
+ \uparrow S_B = \frac{M}{EI}(L) = \frac{ML}{EI}
$$

Ans.

$$
+ \zeta M_B = \frac{M}{EI}(L)\left(\frac{L}{2}\right) = \frac{ML^2}{2EI}
$$

$$
\Delta_B = \frac{ML^2}{2EI}
$$

\n
$$
\Delta_B = \frac{ML^2}{2EI} \uparrow
$$
 Ans.

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Example 6.10

Use the conjugate-beam method to determine the slopes at ends A and D and the deflections at points B and C of the beam shown in Fig. 6.16(a).

Solution

 M/EI Diagram. This beam was analyzed in Example 6.4 by the moment-area method. The M/EI diagram for this beam is shown in Fig. $6.16(b)$.

Conjugate Beam. Fig. 6.16(c) shows the conjugate beam loaded with the M/EI diagram of the real beam. Points A and D , which are simple end supports on the real beam, remain the same on the conjugate beam. Because the M/EI diagram is positive, it is applied as an upward load on the conjugate beam.

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Reactions for Conjugate Beam. By applying the equations of equilibrium to the free body of the entire conjugate beam, we obtain the following:

$$
+\zeta \sum M_D = 0
$$

\n
$$
A_y(40) - \frac{1}{EI} \left[\frac{1}{2} (800) (20) \left(\frac{20}{3} + 20 \right) + 600 (10) (15) + \frac{1}{2} (200) (10) \left(\frac{20}{3} + 10 \right) + \frac{1}{2} (600) (10) \left(\frac{20}{3} \right) \right] = 0
$$

\n
$$
A_y = \frac{8,500 \text{ k} \cdot \text{ft}^2}{EI}
$$
\n
$$
+ \frac{1}{2} \sum F_y = 0
$$
\n
$$
\frac{1}{EI} \left[-8,500 + \frac{1}{2} (800) (20) + 600 (10) + \frac{1}{2} (200) (10) \right] - D_y = 0
$$
\n
$$
D_y = \frac{9,500 \text{ k} \cdot \text{ft}^2}{EI}
$$
\n**Slope at** *A*. The slope at *A* on the real beam is equal to the shear just to the right of $+ \frac{1}{2} S_{AB}$.
\nTherefore, the slope at *A* on the real beam is
\n
$$
\theta_A = \frac{8,500 (12)^2}{EI} = -\frac{8,500 (12)^2}{(1,800) (46,000)} = -0.015
$$
\n
$$
\theta_A = 0.015 \text{ rad}
$$
\n**Slope at** *D*: The slope at *D* on the real beam is equal to the shear just to the left of *I*
\n $+ \frac{1}{2} S_{D,L} = +D_y = \frac{+9,500 (12)^2}{EI}$
\nTherefore, the slope at *D* on the real beam is
\n
$$
\theta_D = \frac{9,500 \text{ k} \cdot \text{ft}^2}{EI} = \frac{9,500 (12)^2}{(1,800) (46,000)} = 0.017 \text{ rad}
$$
\n**Definition at** *B*. The deflection at *B* on the real beam is equal to the bending moment
\nthe free body of the conjugate beam to the left of *B*, we compute
\n $+ \zeta M_B = \frac{1$

$$
\uparrow S_{A,R} = -A_y = -\frac{8,500 \text{ k-fit}^2}{EI}
$$

$$
\theta_A = -\frac{8,500 \text{ k} \cdot \text{ft}^2}{EI} = -\frac{8,500(12)^2}{(1,800)(46,000)} = -0.015 \text{ rad}
$$
\n
$$
\theta_A = 0.015 \text{ rad}
$$
\nAns.

$$
4 \text{ } S_{D,L} = +D_y = \frac{+9,500 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
\theta_D = \frac{9,500 \text{ k-ft}^2}{EI} = \frac{9,500(12)^2}{(1,800)(46,000)} = 0.017 \text{ rad}
$$
\n
$$
\theta_D = 0.017 \text{ rad}
$$
\nAns.

$$
+ \zeta M_B = \frac{1}{EI} \left[-8,500(20) + \frac{1}{2}(800)(20) \left(\frac{20}{3} \right) \right] = -\frac{116,666.67 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\Delta_B = -\frac{116,666.67 \text{ k} \cdot \text{ft}^3}{EI} = -\frac{116,666.67(12)^3}{(1,800)(46,000)} = -2.43 \text{ in.}
$$
\n
$$
\Delta_B = 2.43 \text{ in.} \downarrow \text{Ans.}
$$

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Deflection at C. The deflection at C on the real beam is equal to the bending moment at C in the conjugate beam. Using the free body of the conjugate beam to the right of C , we determine

$$
+ \zeta M_C = \frac{1}{EI} \left[-9,500(10) + \frac{1}{2}(600)(10) \left(\frac{10}{3} \right) \right] = -\frac{85,000 \text{ k} \cdot \text{ft}^3}{EI}
$$

Therefore, the deflection at C on the real beam is

$$
\Delta_C = -\frac{85,000 \text{ k} \cdot \text{ft}^3}{EI} = -\frac{85,000(12)^3}{(1,800)(46,000)} = -1.77 \text{ in.}
$$

\n
$$
\Delta_C = 1.77 \text{ in.}
$$

A_C = 1.77 in. 1
\n**Example 6.11**
\nDetermine the maximum deflection for the beam shown in Fig. 6.17(a) by the conjugate-beam n
\n**Solution**
\n*M[E1* Diagram. This beam was previously analyzed in Example 6.5 by the moment-area method for the beam is shown in Fig. 6.17(b).
\nConjugate Bama. The simply supported conjugate beam, loaded with the *M/E1* diagram of the
\nFig. 6.17(c).
\n**Reaction at Support A of the Conjugate beam.** By applying the moment equilibrium equation
\nbody of the entire conjugate beam, we determine
\n
$$
+ \zeta Mc = 0
$$

\n $A_y(15) - \frac{1}{EI} [\frac{1}{2} (400)(10)(\frac{10}{3} + 5) + \frac{1}{2} (400)(5)(\frac{10}{3})] = 0$
\n $A_y = \frac{1.333.33 \text{ kV} \cdot \text{m}^2}{EI}$
\n**Location of the Maximum Bending Moment in Conjugate beam.** If the maximum bending moment
\nFig. 6.17(c)), then the shear in the conjugate beam at *D* must be zero. Considering the free body of the cell beam
\nthe left of *D*, we write
\n $+1 S_D = \frac{1}{EI} [-1,333.33 + \frac{1}{2} (40x_m)(x_m)] = 0$
\nfrom which
\n $x_m = 8.16 \text{ m}$
\nMaximum Deflection of the Real Bean. The maximum deflection of the real beam is equal to
\nmoment in the conjugate beam, which can be determined by considering the free body of the cor-

+
$$
\uparrow S_D = \frac{1}{EI} \left[-1,333.33 + \frac{1}{2} (40x_m)(x_m) \right] = 0
$$

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$$
+ \zeta M_{\text{max}} = M_D = \frac{1}{EI} \left[-1,333.33(8.16) + \frac{1}{2} (40)(8.16)^2 \left(\frac{8.16}{3} \right) \right]
$$

$$
= -\frac{7,244.51 \text{ kN} \cdot \text{m}^3}{EI}
$$

continued

Ans.

Example 6.12

Determine the slope at point A and the deflection at point C of the beam shown in Fig. 6.18(a) by the conjugate-beam method.

points D and E, which are simple interior supports on the real beam, become internal hinges on the conjugate beam; point C, which is an internal hinge on the real beam, becomes a simple interior support on the conjugate beam. Also note that the positive part of the M/EI diagram is applied as upward loading on the conjugate beam, whereas the negative part of the M/EI diagram is applied as downward loading.

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Reaction at Support A of the Conjugate Beam. We determine the reaction A_y of the conjugate beam by applying the equations of condition as follows:

$$
+ \zeta \sum M_D^{AD} = 0
$$

$$
A_y(30) - \frac{1}{2} \left(\frac{100}{EI} \right) (20)(20) + C_y(10) + \frac{1}{2} \left(\frac{200}{EI} \right) (10) \left(\frac{10}{3} \right) = 0
$$

 or

+
$$
\zeta \sum M_{E}^{AE} = 0
$$

\n $A_y(45) - \frac{1}{2} \left(\frac{100}{EI} \right) (20)(35) + C_y(25) + \frac{1}{2} \left(\frac{200}{EI} \right) (10) \left(\frac{10}{3} + 15 \right)$
\n+ $\frac{150}{EI} (15)(7.5) + \frac{1}{2} \left(\frac{50}{EI} \right) (15)(10) = 0$
\nor
\n $45A_y + 25C_y = -\frac{3,958.33}{EI}$
\nSubstituting Eq. (1) into Eq. (2) and solving for A_y , we obtain
\n $A_y = \frac{1,520.83 \text{ k} \cdot \text{R}^2}{EI}$
\nSlope at A. The slope at A on the real beam is equal to the shear just to the right of A in the con-
\n+ $15A_x = -A_y = -\frac{1,520.83 \text{ k} \cdot \text{R}^2}{EI}$
\nTherefore, the slope at A on the real beam is
\n $\theta_A = -\frac{1,520.83}{EI} = -\frac{1,520.83(12)^2}{(29,000)(2,500)} = -0.003 \text{ rad}$
\n $\theta_A = 0.003 \text{ rad}$
\n $\theta_A = 0.003$

 $C_y = -3A_y + \frac{1,666.67}{EI}$

 (2)

 (1)

$$
+ \uparrow S_{A,R} = -A_y = -\frac{1,520.83 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
\theta_A = -\frac{1,520.83}{EI} = -\frac{1,520.83(12)^2}{(29,000)(2,500)} = -0.003 \text{ rad}
$$

\n
$$
\theta_A = 0.003 \text{ rad}
$$

$$
+\zeta M_C = \frac{1}{EI} \left[-1{,}520.83(20) + \frac{1}{2}(100)(20)(10) \right] = -\frac{20{,}416.67 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\Delta_C = -\frac{20,416.67 \text{ k} \cdot \text{ft}^3}{EI} = -\frac{20,416.67(12)^3}{(29,000)(2,500)} = -0.487 \text{ in.}
$$

\n
$$
\Delta_C = 0.487 \text{ in.} 1
$$

Solution

 $M|E$ I Diagram. This beam was previously analyzed in Example 6.7 by the moment-area method. The M/EI diagram by cantilever parts with respect to point B is shown in Fig. 6.19(b).

 $\frac{1}{\lambda}$

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continued

Ans.

$$
+ \zeta \sum M_B^{AB} = 0
$$

$$
A_y(30) + \frac{1}{EI} \left[\frac{1}{3} (900)(30) \left(\frac{30}{4} \right) - \frac{1}{2} (780)(30) \left(\frac{30}{3} \right) \right] = 0
$$

$$
A_y = \frac{1,650 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
+ \zeta M_C = \frac{1}{EI} \left[-1,650(40) - \frac{1}{3}(900)(30) \left(\frac{30}{4} + 10 \right) + \frac{1}{2}(780)(30)(20) - \frac{1}{2}(120)(10) \left(\frac{20}{3} \right) \right] = \frac{6,500 \text{ k} \cdot \text{ft}^3}{EI}
$$

Therefore, the deflection at C on the real beam is

$$
\Delta_C = \frac{6,500 \text{ k} \cdot \text{ft}^3}{EI} = \frac{6,500(12)^3}{(29,000)(2,000)} = 0.194 \text{ in}
$$

$$
\Delta_C = 0.194 \text{ in.} \uparrow
$$

Ans.

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Summary

In this chapter we have discussed the geometric methods for determining the slopes and deflections of statically determinate beams. The differential equation for the deflection of beams can be expressed as

$$
\frac{d^2y}{dx^2} = \frac{M}{EI} \tag{6.9}
$$

 \mathbf{B} \mathbf{A}

First moment-area theorem:
$$
\theta_{BA} = \int_{A}^{B} \frac{M}{EI} dx
$$
 (6.12)

 (6.15)

The direct integration method exertistics contained as such
successively to obtain equations for the such successively to obtain equations for the
curve. The constants of integration as is considered to several location
of

6.1 through 6.6 Determine the equations for slope and deflection of the beam shown by the direct integration method. $EI = constant$.

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Example 6.1

Determine the equations for the slope and deflection of the beam shown in Fig. 6.2(a) by the direct integration method. Also, compute the slope at each end and the deflection at the midspan of the beam. EI is constant.

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$$
M = \frac{wL}{2}(x) - (wx)\left(\frac{x}{2}\right) = \frac{w}{2}(Lx - x^2)
$$

Equation for $M|EL$. The flexural rigidity, EI, of the beam is constant, so the equation for M/EL can be written as

$$
\frac{d^2y}{dx^2} = \frac{M}{EI} = \frac{w}{2EI}(Lx - x^2)
$$

continual

Equations for Slope and Deflection. The equation for the slope of the elastic curve of the beam can be obtained by integrating the equation for M/EI as

$$
\theta = \frac{dy}{dx} = \frac{w}{2EI} \left(\frac{Lx^2}{2} - \frac{x^3}{3} \right) + C_1
$$

Integrating once more, we obtain the equation for deflection as

$$
y = \frac{w}{2EI} \left(\frac{Lx^3}{6} - \frac{x^4}{12} \right) + C_1 x + C_2
$$

At end A,
$$
x = 0
$$
, $y = 0$
At end B, $x = L$, $y = 0$

The constants of inegrators, C_1 and C_2 , are countants of supposite interesting
 A at end B , $x = L$, $y = 0$

By applying the first boundary condition—that is, by setting $x = L$ and $y = 0$

Next, by using the seco

$$
0 = \frac{w}{2EI} \left(\frac{L^4}{6} - \frac{L^4}{12} \right) + C_1
$$

$$
= \frac{w}{2EI} \left(\frac{Lx^2}{2} - \frac{x^3}{3} - \frac{L^3}{12} \right)
$$
 (1) Ans

$$
u = \frac{wx}{12EI} \left(Lx^2 - \frac{x^3}{2} - \frac{L^3}{2} \right)
$$
 (2) Ans.

$$
\theta_A = -\frac{wL^3}{24EI}
$$
 or $\theta_A = \frac{wL^3}{24EI}$ $\sqrt{ }$ Ans.

$$
\theta_B = \frac{wL^3}{24EI} \quad \text{or} \quad \theta_B = \frac{wL^3}{24EI} \quad \overline{\smash{\bigtriangledown}}
$$

$$
C = -\frac{5wL^4}{384EI} \text{ or } y_C = \frac{5wL^4}{384EI} \downarrow
$$
 Ans.

Solution

Equation for Bending Moment. We pass a section at a distance x from support A , as shown in Fig. 6.3(b). Considering the free body to the right of this section, we write the equation for bending moment as

$$
M = -15(20 - x)
$$

continued

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$$
\frac{d^2y}{dx^2} = \frac{M}{EI} = -\frac{15}{EI}(20 - x)
$$

$$
\theta = \frac{dy}{dx} = -\frac{15}{EI} \left(20x - \frac{x^2}{2} \right) + C_1
$$

$$
y = -\frac{15}{EI} \left(10x^2 - \frac{x^3}{6} \right) + C_1 x + C_2
$$

$$
\theta = -\frac{15}{EI} \left(20x - \frac{x^2}{2} \right)
$$

$$
y = -\frac{15}{EI} \left(10x^2 - \frac{x^3}{6} \right)
$$

Slope and Deflection at End B. By substituting $x = 20$ ft, $E = 29,000(12^2)$ ksf, and $I = 758/(12^4)$ ft⁴ into the foregoing equations for slope and deflection, we obtain

$$
\theta_B = -0.0197 \text{ rad}
$$
 or $\theta_B = 0.0197 \text{ rad}$ Ans.
\n $y_B = -0.262 \text{ ft} = -3.14 \text{ in.}$ or $y_B = 3.14 \text{ in.}$ Ans.

Example 6.3

and deflections at points *B* and *C* of the cantilever beam shown in Fig. 6
 Joanna Prices of the contract diagram for the beam is shown in Fig. 6.5(b)
 Joseph Contract diagram for the beam is shown in Fig. 6.5(b) method.

Solution

confirmed

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$$
I_{AB} = 6,000 = 2(3,000) = 2I
$$

$$
\theta_{BA} = \frac{1}{EI} \left[(100)(15) + \frac{1}{2} (150)(15) \right] = \frac{2{,}625 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
B = \theta_{BA} = \frac{2.625 \text{ k} \cdot \text{ft}^2}{EI} = \frac{2.625(12)^2 \text{ k} \cdot \text{in.}^2}{EI}
$$

$$
\theta_B = \frac{2{,}625(12)^2}{(29,000)(3,000)}
$$
 rad = 0.0043 rad
\n
$$
\theta_B = 0.0043
$$
 rad

Deflection at B . From Fig. 6.5(d), it can be seen that the deflection of B with respect to the undeformed axis of the beam is equal to the tangential deviation of B from the tangent at A ; that is,

 $\Delta_B = \Delta_{BA}$

According to the second moment-area theorem,

$$
\Delta_{BA} =
$$
 moment of the area of the *M/EI* diagram between *A* and *B* about *B*

$$
\Delta_B = \Delta_{BA} = \frac{22,500 \text{ k-ft}^3}{EI}
$$

=
$$
\frac{22,500(12)^3}{(29,000)(3,000)} = 0.45 \text{ in.}
$$

$$
\Delta_B = 0.45 \text{ in.}
$$

$$
=\frac{1}{EI}\left[(100)(15) + \frac{1}{2}(150)(15) + \frac{1}{2}(200)(10) \right] = \frac{3,625 \text{ k·ft}}{EI}
$$

Therefore,

\n
$$
\Delta_B = \Delta_{BA} = \frac{22,500 \text{ k} \text{ ft}^3}{EI}
$$
\n
$$
= \frac{22,500 \text{ (12)}^3}{(29,000)(3,000)} = 0.45 \text{ in.}
$$
\n
$$
\Delta_B = 0.45 \text{ in.}
$$
\nSince at C. From Fig. 6.5(d), we can see that

\n
$$
\theta_{C4} = \text{area of the } M/EI \text{ diagram between } A \text{ a}
$$
\n
$$
= \frac{1}{EI} \left[(100)(15) + \frac{1}{2}(150)(15) + \frac{1}{2}(200)(1000) \right]
$$
\nTherefore,

\n
$$
\theta_C = \theta_{C4} = \frac{3,625 \text{ k} \cdot \text{ft}^2}{EI}
$$
\n
$$
= \frac{3,625(12)^2}{(29,000)(3,000)} = 0.006 \text{ rad}
$$
\n
$$
\theta_C = 0.006 \text{ rad}
$$
\n
$$
\theta_C = 0.006 \text{ rad}
$$
\nDeflection at C. It can be seen from Fig. 6.5(d) that

\n
$$
\Delta_{C4} = \text{moment of the area of the } M/EI \text{ diagram between } A \text{ and } B \text{ is } \Delta_{C4} = \Delta_{C4} \text{ where}
$$
\n
$$
\Delta_{C4} = \text{moment of the area of the } M/EI \text{ diagram between } A \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta_{C4} = \text{ moment of the } B \text{ and } B \text{ is } \Delta
$$

Ans.

Ans.

Ans.

$$
= \frac{1}{EI} \left[(100)(15)(7.5 + 10) + \frac{1}{2}(150)(15)(10 + 10) + \frac{1}{2}(200)(10)(6.67) \right]
$$

=
$$
\frac{55,420 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\Delta c = \Delta c_A = \frac{55,420 \text{ k} \cdot \text{ft}^3}{EI}
$$

=
$$
\frac{55,420(12)^3}{(29,000)(3,000)} = 1.1 \text{ in.}
$$

$$
\Delta c = 1.1 \text{ in.} \downarrow
$$

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Example 6.4

Use the moment-area method to determine the slopes at ends A and D and the deflections at points B and C of the beam shown in Fig. 6.6(a).

Solution

 M/EI Diagram. Because EI is constant along the length of the beam, the shape of the M/EI diagram is the same as that of the bending moment diagram. The M/EI diagram is shown in Fig. 6.6(b).

Slope at A. The slope of the elastic curve is not known at any point on the beam, so we will use the tangent at support A as the reference tangent and determine its slope, θ_d , from the conditions that the deflections at the support points A and D are zero. From Fig. $6.6(c)$, we can see that

$$
\theta_A = \frac{\Delta_{DA}}{L}
$$

in which θ_A is assumed to be so small that tan $\theta_A \approx \theta_A$. To evaluate the tangential deviation Δ_{DA} , we apply the second moment-area theorem:

 Δ_{DA} = moment of the area of the *M/EI* diagram between *A* and *D* about *D*

$$
\Delta_{DA} = \frac{1}{EI} \left[\frac{1}{2} (800)(20) \left(\frac{20}{3} + 20 \right) + \frac{1}{2} (200)(10) \left(\frac{20}{3} + 10 \right) \right]
$$

\n
$$
+ 600(10)(15) + \frac{1}{2} (600)(10) \left(\frac{20}{3} \right) \right]
$$

\n
$$
= \frac{340,000 \text{ k} \cdot \text{ft}^3}{EI}
$$

\nTherefore, the slope at *A* is
\n
$$
\theta_A = \frac{\Delta_{DA}}{L} = \frac{340,000/EI}{40} = \frac{8,500 \text{ k} \cdot \text{ft}^2}{EI}
$$

\nSubstituting the numerical values of *E* and *I*, we obtain
\n
$$
\theta_A = \frac{8,500(12)^2}{(1,800)(46,000)} = 0.015 \text{ rad}
$$

\n
$$
\theta_A = 0.015 \text{ rad}
$$

\n
$$
\theta_B = \theta_{DA} - \theta_A
$$

\nin which, according to the first moment-area theorem,
\n
$$
\theta_{DA} = \text{area of the } M/EI
$$
 diagram between *A* and
\n
$$
= \frac{1}{EI} \left[\frac{1}{2} (800)(20) + \frac{1}{2} (200)(10) + 600(10) + \frac{1}{2} \right]
$$

\n
$$
= \frac{18,000 \text{ k} \cdot \text{ft}^2}{EI}
$$

\nTherefore,
\n
$$
\theta_D = \frac{18,000}{EI} - \frac{8,500}{EI} = \frac{9,500 \text{ k} \cdot \text{ft}^2}{EI}
$$

\n
$$
\theta_D = \frac{9,500(12)^2}{(1,800)(46,000)} = 0.017 \text{ rad}
$$

\n
$$
\theta_D = 0.017 \text{ rad}
$$

\n
$$
\theta_D = 0.017 \text{ rad}
$$

\n
$$
\theta_D = 0.017 \text{ rad}
$$

\n
$$
\
$$

$$
\theta_A = \frac{\Delta_{DA}}{L} = \frac{340,000/EI}{40} = \frac{8,500 \text{ k} \cdot \text{ft}^2}{EI}
$$

$$
\theta_A = \frac{8,500(12)^2}{(1,800)(46,000)} = 0.015 \text{ rad}
$$

$$
\theta_A = 0.015 \text{ rad}
$$

$$
\theta_D = \theta_{DA} - \theta_A
$$

$$
= \frac{1}{EI} \left[\frac{1}{2} (800)(20) + \frac{1}{2} (200)(10) + 600(10) + \frac{1}{2} (600)(10) \right]
$$

=
$$
\frac{18,000 \text{ k-fr}}{EI}
$$

$$
\theta_D = \frac{18,000}{EI} - \frac{8,500}{EI} = \frac{9,500 \text{ k} \cdot \text{ft}^2}{EI}
$$

\n
$$
\theta_D = \frac{9,500(12)^2}{(1,800)(46,000)} = 0.017 \text{ rad}
$$

\n
$$
\theta_D = 0.017 \text{ rad}
$$

$$
\theta_A = \frac{\Delta_B + \Delta_{BA}}{20}
$$

from which

₹

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$$
\Delta_B = 20\theta_A - \Delta_{BA}
$$

continued

Ans.

where

 Δ_{B4} = moment of the area of the M/EI diagram between A and B about B

$$
= \frac{1}{EI} \left[\frac{1}{2} (800)(20) \left(\frac{20}{3} \right) \right]
$$

$$
= \frac{53,333.33 \text{ k} - \text{ft}^3}{EI}
$$

Therefore,

$$
\Delta_B = 20 \left(\frac{8,500}{EI}\right) - \frac{53,333.33}{EI} = \frac{116,666.67 \text{ k} \text{A}^3}{EI}
$$

\n
$$
\Delta_B = \frac{116,666.67(12)^3}{(1,800)(46,000)} = 2.43 \text{ in.}
$$

\n
$$
\Delta_B = 2.43 \text{ in.} \perp
$$

\nDeflection at C. Finally, considering the portion CD of the elastic curve in Fig. 6.6(c) and assuming θ_D to be small (so that $\tan \theta_D \approx \theta_D$), we write
\n
$$
\theta_D = \frac{\Delta_C + \Delta_{CD}}{10}
$$

\nor
\nwhere
\n
$$
\Delta_{CD} = \frac{1}{EI} \left[\frac{1}{2} (600)(10) \left(\frac{10}{3} \right) \right] = \frac{10,000 \text{ k} \cdot \text{A}^3}{EI}
$$

\n
$$
\Delta_C = 10 \left(\frac{9,500}{EI} \right) - \frac{10,000}{EI} = \frac{85,000 \text{ k} \cdot \text{A}^3}{EI}
$$

\n
$$
\Delta_C = \frac{85,000(12)^3}{(1,800)(46,000)} = 1.77 \text{ in.}
$$

\n
$$
\Delta_C = 1.77 \text{ in.} \downarrow
$$
 Ans.
\n**Example 6.5**
\nDetermine the maximum deflection for the beam shown in Fig. 6.7(a) by the momentarca method.

$$
\Delta_C = 10 \left(\frac{9,500}{EI} \right) - \frac{10,000}{EI} = \frac{85,000 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\Delta_C = \frac{85,000(12)^3}{(1,800)(46,000)} = 1.77 \text{ in.}
$$

$$
\Delta_C = 1.77 \text{ in.}
$$

Ans.

continued

Solution

 M/EI Diagram. The M/EI diagram is shown in Fig. 6.7(b).

Elastic Curve. The elastic curve for the beam is shown in Fig. $6.7(c)$.

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$$
\theta_A = \frac{\Delta_{CA}}{15}
$$

$$
\Delta_{Cd} = \frac{1}{EI} \left[\frac{1}{2} (400)(10) \left(\frac{10}{3} + 5 \right) + \frac{1}{2} (400)(5) \left(\frac{10}{3} \right) \right]
$$

=
$$
\frac{20,000 \text{ kN} \cdot \text{m}^3}{EI}
$$

Therefore, the slope at A is

$$
\lambda_t = \frac{20,000/EI}{15} = \frac{1,333,33 \text{ kN} \cdot \text{m}^2}{EI}
$$

Location of the Maximum Deflection. If the maximum deflection occurs at point D , located at a distance x_m from the left support A (see Fig. 6.7(c)), then the slope at D must be zero; therefore,

$$
\theta_{DA} = \theta_A = \frac{1,333.33 \text{ kN} \cdot \text{m}^2}{EI}
$$

$$
\theta_{DA}
$$
 = area of the $\frac{M}{EI}$ diagram between A and $D = \frac{1,333.33}{EI}$

$$
x_m = 8.16 \text{ m}
$$

Jozvebama.ir

$$
_{\text{max}} = \frac{7,244.51 \text{ kN} \cdot \text{m}^3}{EI}
$$

$$
\Delta_{\text{max}} = \frac{7,244.51}{200(10^6)(700)(10^{-6})} = 0.0517 \text{ m}
$$

$$
\Delta_{\text{max}} = 51.7 \text{ mm} \downarrow
$$

Ans.

continued

Use the moment-area method to determine the slope at point A and the deflection at point C of the beam shown in Fig. 6.8(a).

Solution

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 M/EI Diagram. The bending moment diagram is shown in Fig. 6.8(b), and the M/EI diagram for a reference moment of inertia $I = 2,500$ in.⁴ is shown in Fig. 6.8(c).

$$
\theta_D = \frac{\Delta_{ED}}{15}
$$

$$
\Delta_{ED} = \frac{1}{EI} \left[150(15)(7.5) + \frac{1}{2}(50)(15)(10) \right] = \frac{20,625 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\theta_D = \frac{20,625}{15(EI)} = \frac{1,375 \text{ k} \cdot \text{ft}^2}{EI}
$$

in which

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$$
\Delta_{CD} = \frac{1}{2} \left(\frac{200}{EI} \right) (10) \left(\frac{20}{3} \right) = \frac{6,666.67 \text{ k} \cdot \text{ft}^3}{EI}
$$
Therefore,

$$
\Delta_C = 10 \left(\frac{1,375}{EI} \right) + \frac{6,666.67}{EI} = \frac{20,416.67 \text{ k} \cdot \text{ft}^3}{EI}
$$

$$
\Delta_C = \frac{20,416.67(12)^3}{(29,000)(2,500)} = 0.487
$$
 in.
\n
$$
\Delta_C = 0.487
$$
 in. \downarrow Ans.

$$
\theta_d = \frac{\Delta_C + \Delta_{Cd}}{20}
$$

where

$$
\Delta_{Cd} = \frac{1}{2} \left(\frac{100}{EI} \right) (20) (10) = \frac{10,000 \text{ k-fit}^3}{EI}
$$

Therefore,

e numerical values of E and I, we obtain
\n
$$
\Delta_C = \frac{20,416.67(12)^3}{(29,000)(2,500)} = 0.487 \text{ in.}
$$
\n
$$
\Delta_C = 0.487 \text{ in.} \downarrow
$$
\nconsidering the portion AC of the elastic curve, we can see from Fig. 6.8(d) that
\n
$$
\theta_4 = \frac{\Delta_C + \Delta_{C4}}{20}
$$
\n
$$
\Delta_{CA} = \frac{1}{2} \left(\frac{100}{EI}\right) (20) (10) = \frac{10,000 \text{ k} \cdot \text{ft}^3}{EI}
$$
\n
$$
\theta_4 = \frac{1}{20} \left(\frac{20,416,67}{EI} + \frac{10,000}{EI}\right) = \frac{1,520.83 \text{ k} \cdot \text{ft}^2}{EI}
$$
\n
$$
\theta_4 = \frac{1,520.83(12)^2}{(29,000)(2,500)} = 0.003 \text{ rad}
$$
\n
$$
\theta_4 = 0.003 \text{ rad}
$$

Ans.

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Example 6.7

Determine the deflection at point C of the beam shown in Fig. 6.11(a) by the moment-area method.

determined in Fig. 6.10. The ordinates of the bending moment diagram are divided by EI to obtain the M/EI diagram shown in Fig. 6.11(b).

continued

Elastic Curve. See Fig. 6.11(c).

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Slope at B . Selecting the tangent at B as the reference tangent, it can be seen from Fig. 6.11(c) that

$$
\theta_B = \frac{\Delta_{AB}}{30}
$$

$$
M/EI \text{ diagram (Fig. 6.11(b)) and the properties of geometric shapes given in Appendix}
$$
\n
$$
\Delta_{AB} = \frac{1}{EI} \left[\frac{1}{2} (780)(30)(20) - \frac{1}{3} (900)(30) \left(\frac{3}{4} \right) (30) \right]
$$
\n
$$
= \frac{31,500 \text{ k} \cdot \text{ft}^3}{EI}
$$
\n
$$
\theta_B = \frac{31,500}{30EI} = \frac{1,050 \text{ k} \cdot \text{ft}^2}{EI}
$$
\n
$$
\Delta_C = 10\theta_B - \Delta_{CB}
$$
\n
$$
\Delta_{CE} = \frac{1}{2} \left(\frac{120}{EI} \right) (10) \left(\frac{20}{3} \right) = \frac{4,000 \text{ k} \cdot \text{ft}^3}{EI}
$$
\n
$$
\Delta_C = 10 \left(\frac{1,050}{EI} \right) - \frac{4,000}{EI} = \frac{6,500 \text{ k} \cdot \text{ft}^3}{EI}
$$
\nthe numerical values of *E* and *I*, we obtain\n
$$
\Delta_C = \frac{6,500(12)^3}{(29,000)(2,000)} = 0.194 \text{ in.}
$$
\n
$$
\Delta_C = 0.194 \text{ in.} \uparrow
$$

Therefore,

$$
\theta_B = \frac{31,500}{30EI} = \frac{1,050 \text{ k-ft}^2}{EI}
$$

Deflection at
$$
C
$$
. From Fig. 6.11(c), we can see that

$$
\Delta_C = 10\theta_B - \Delta_{CB}
$$

where

$$
\Delta_{CB} = \frac{1}{2} \left(\frac{120}{EI} \right) (10) \left(\frac{20}{3} \right) = \frac{4,000 \text{ k-ft}}{EI}
$$

Therefore,

$$
\Delta_C = 10 \left(\frac{1,050}{EI} \right) - \frac{4,000}{EI} = \frac{6,500 \text{ k·ft}}{EI}
$$

$$
\Delta_C = \frac{6,500(12)^3}{(29,000)(2,000)} = 0.194
$$
in.

$$
\Delta_C = 0.194
$$
in. †

Ans.

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